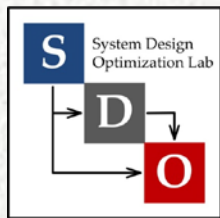


Enabling All-Access Mobility for Planetary Exploration Vehicles via Transformative Reconfiguration



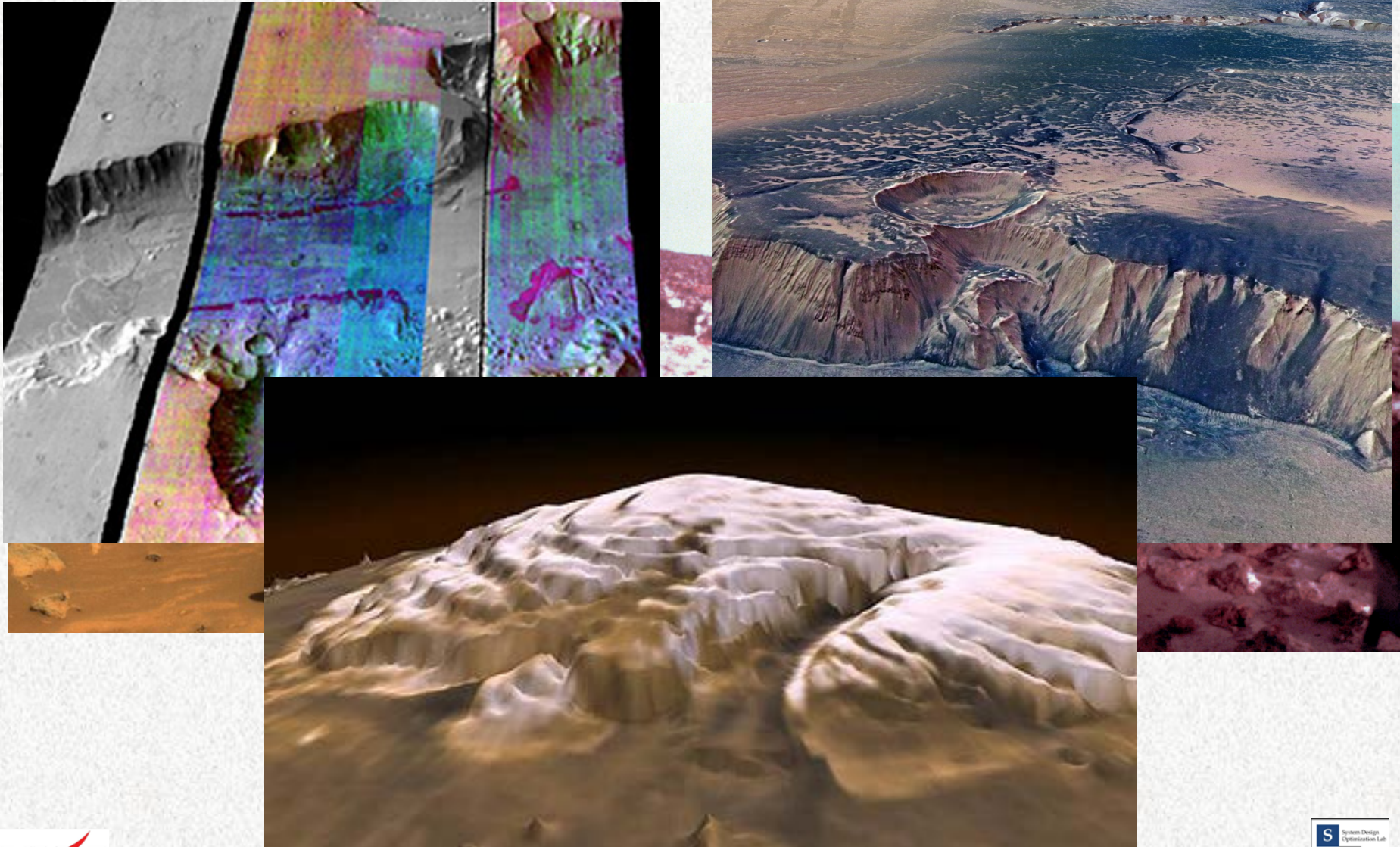
Scott Ferguson
Andre Mazzoleni
3 graduate students
19 undergraduate students
Mechanical and Aerospace Engineering
North Carolina State University



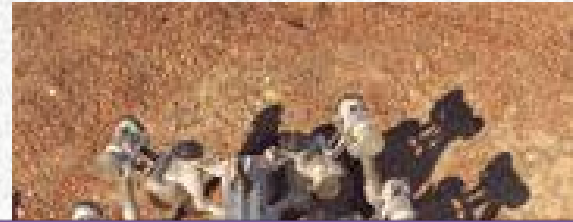
We thank NIAC for their support of this work



Challenges of chaotic terrain



Is mobility enough?



How can we encourage concepts that are even more different?



What potential advantages / mission types could we then do?

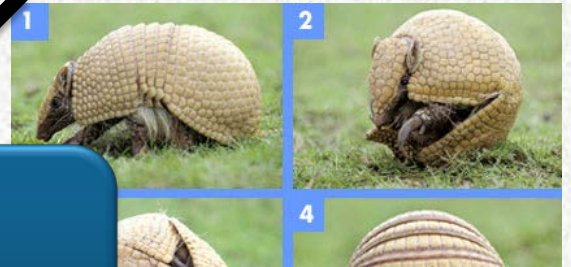


Four drivers of reconfigurability

Expand / Collapse

Expose / Cover

Multi-ability



Reorientation

Fuse / Divide

The need to constrain reconfigurability



Project objective



Multiple scientists wanting to:

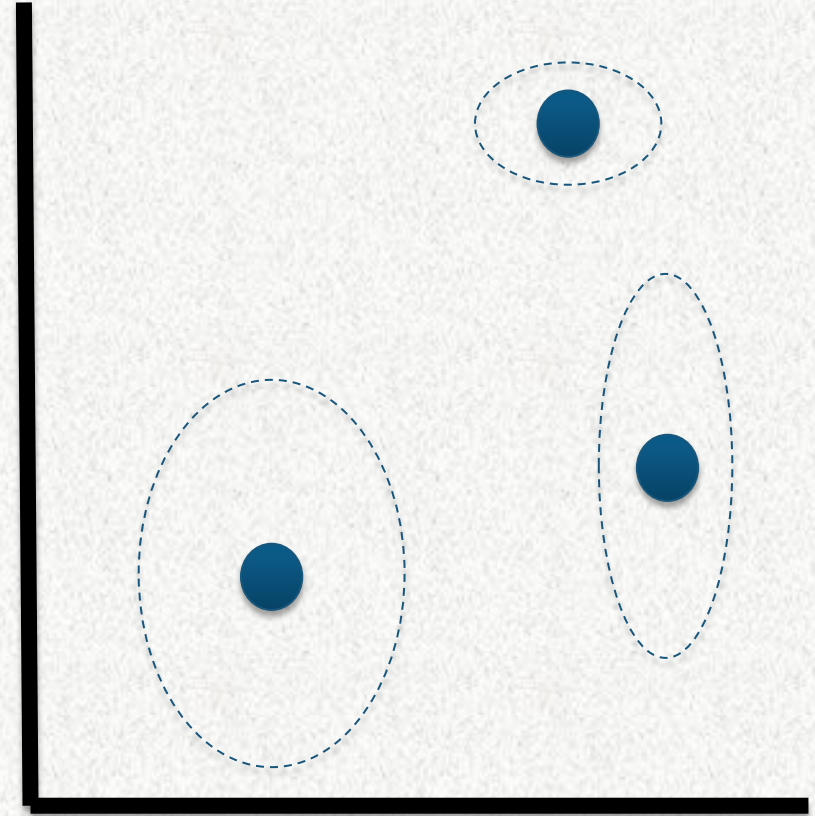
- Sense something
- Measure something
- Explore something



Engineers want something:

- Low risk
- Easily analyzable
- Realizable

Project objective



What makes for an effective configuration??

Leveraging the university infrastructure

3 Graduate students
(Mars exploration class)

19 Undergraduate students
(Capstone design)

Identify mission, science, and requirements

Develop conceptual solutions

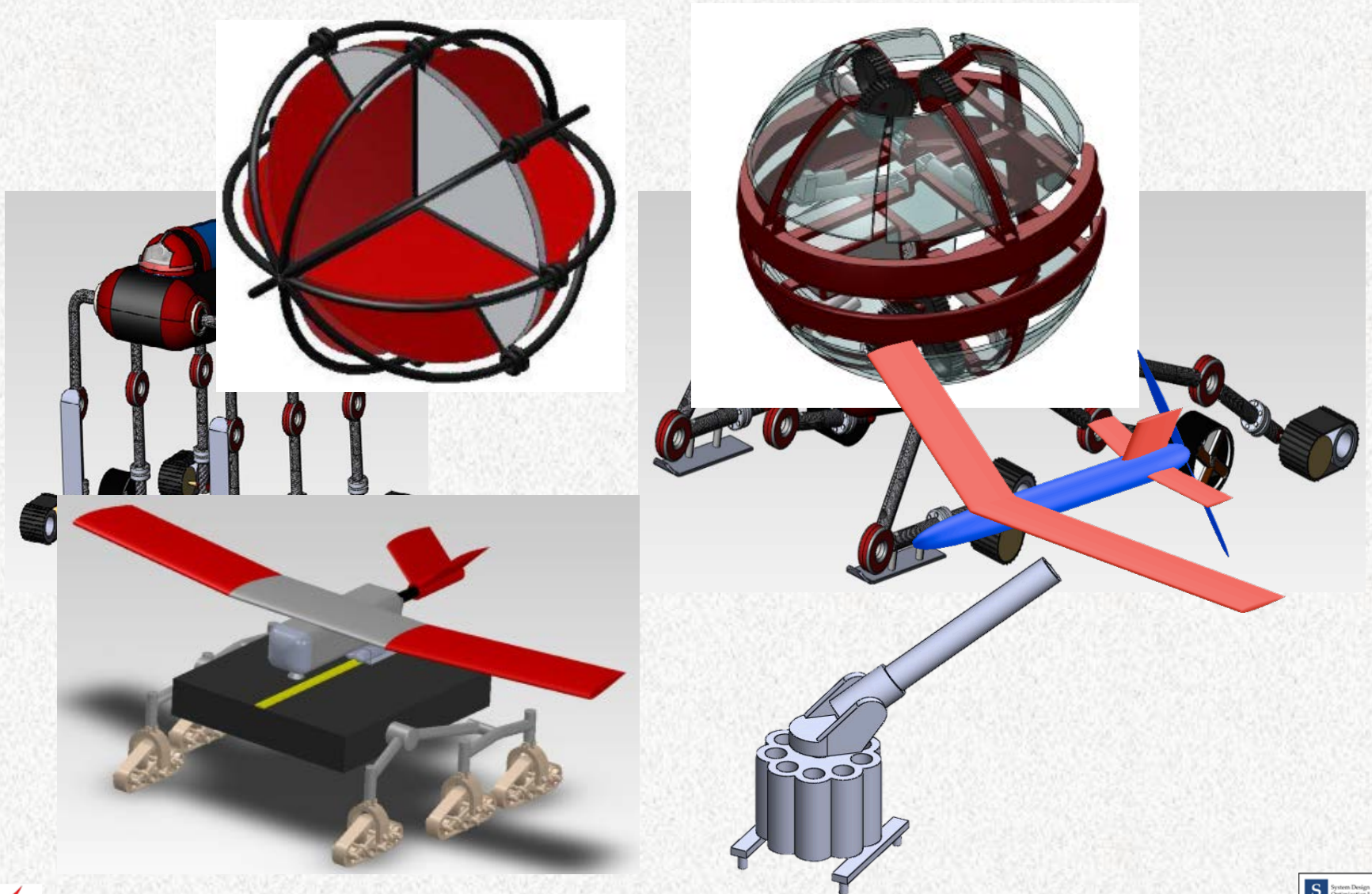
Conduct analysis

Initial prototyping

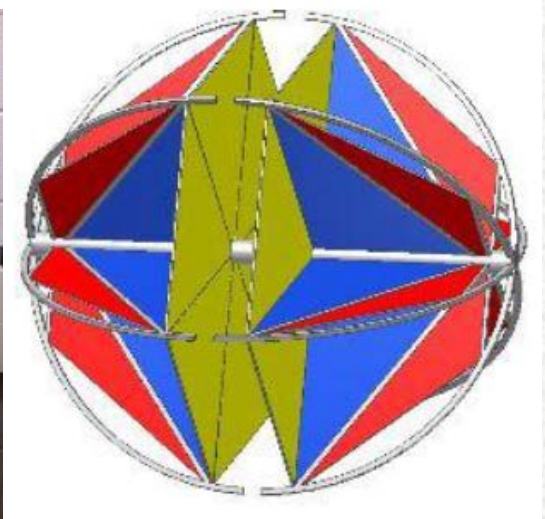
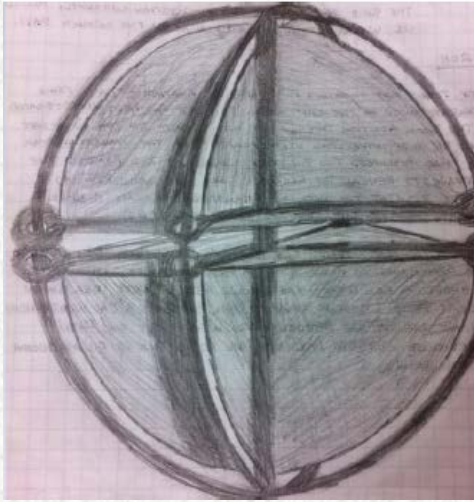
Tradespace exploration

Advanced simulation

What baselines have been created so far?

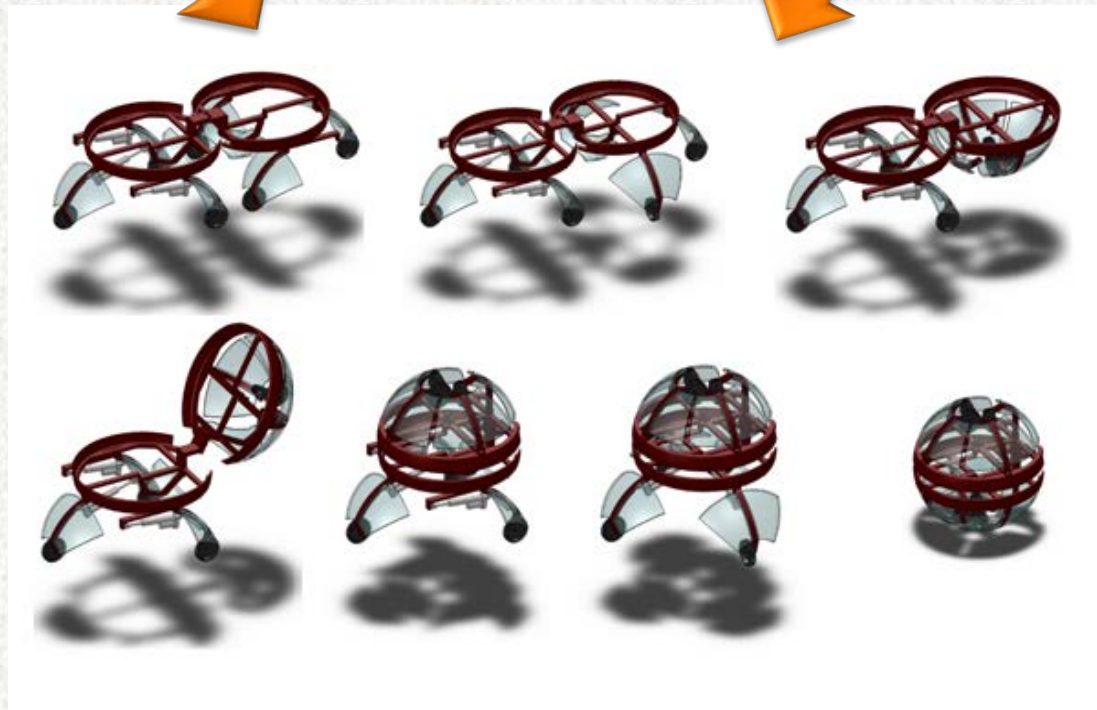


Development of the tumbleweed rover



$$h_{max} = r \left\{ 1 - \sqrt{\frac{C_D}{C_D + 1}} \right\}$$

Inspiration for the TRRex

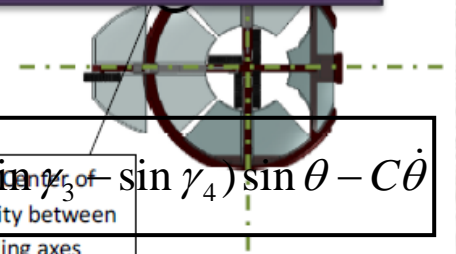
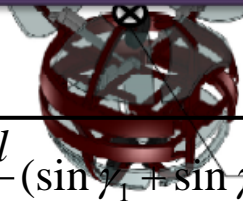
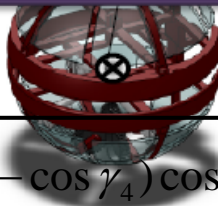


Modeling / analysis of rolling motion

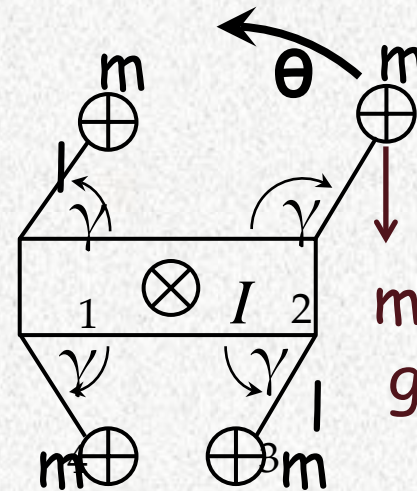
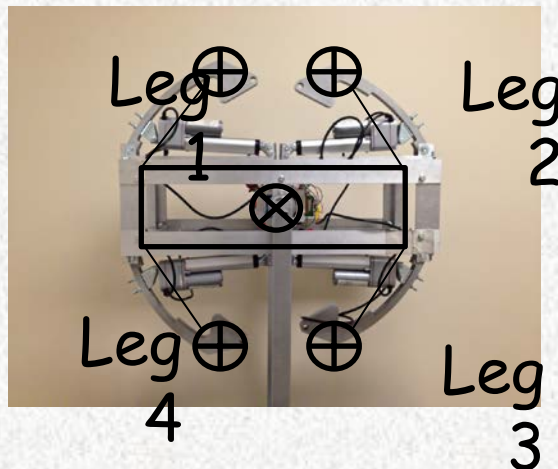
Dynamic modeling of a single DOF system
(as a lumped parameter system)

us rolling in the
presence of a

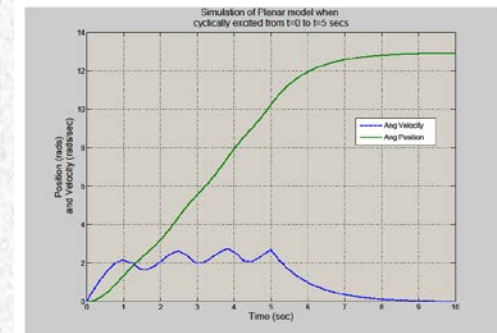
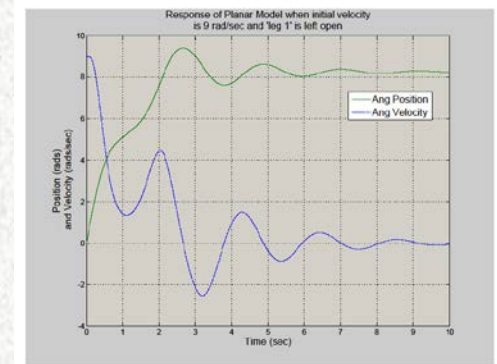
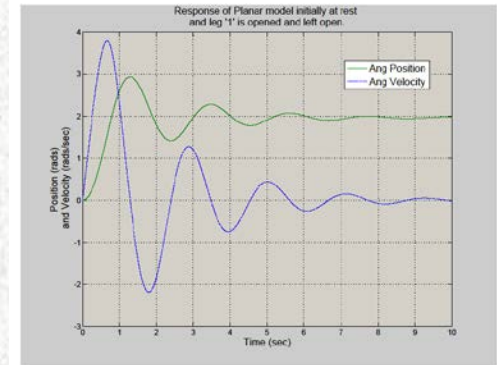
$$\ddot{\theta} = \frac{mgl}{I} (-\cos \gamma_1 + \cos \gamma_2 + \cos \gamma_3 - \cos \gamma_4) \cos \theta + \frac{mgl}{I} (\sin \gamma_1 + \sin \gamma_2 - \sin \gamma_3 - \sin \gamma_4) \sin \theta - C\dot{\theta}$$



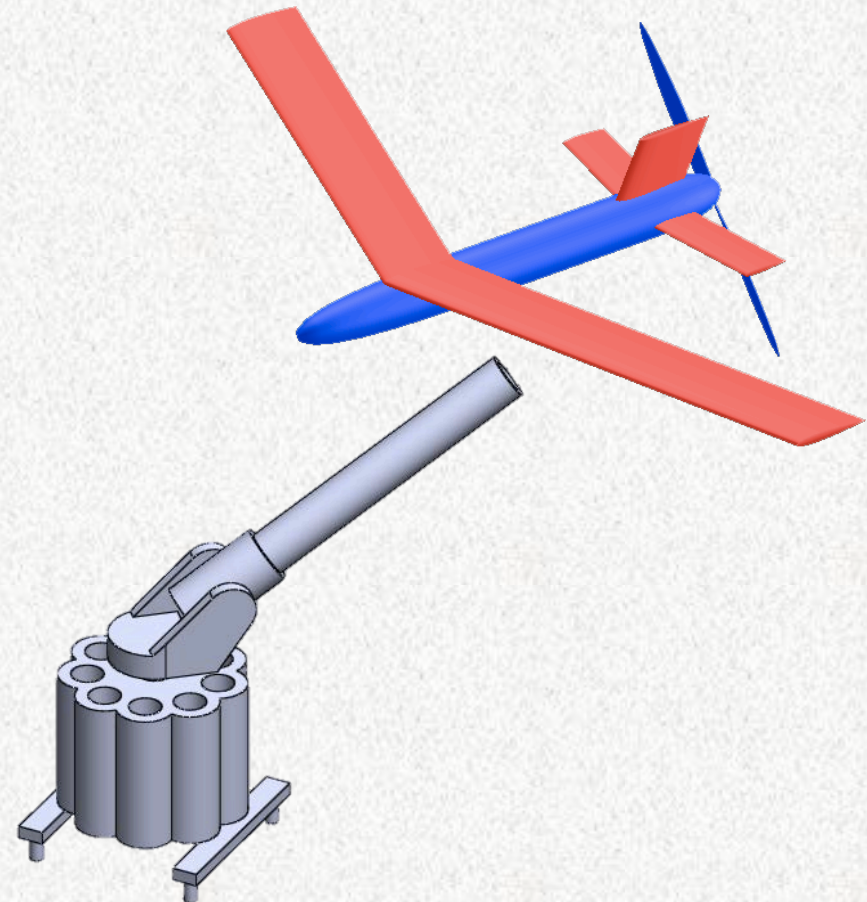
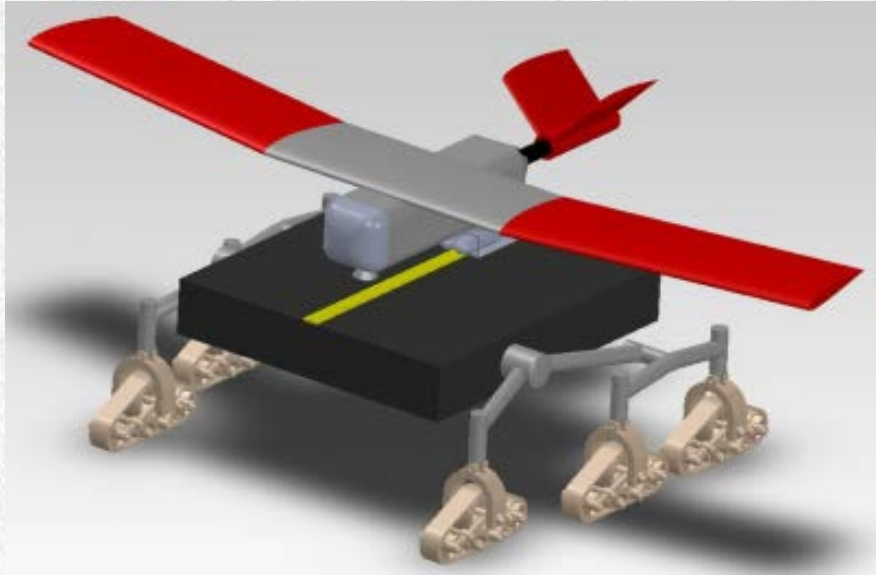
Net torque of Gravity between 2 rolling axes



Prototype and testing



Air-based concepts



Melas Chasma

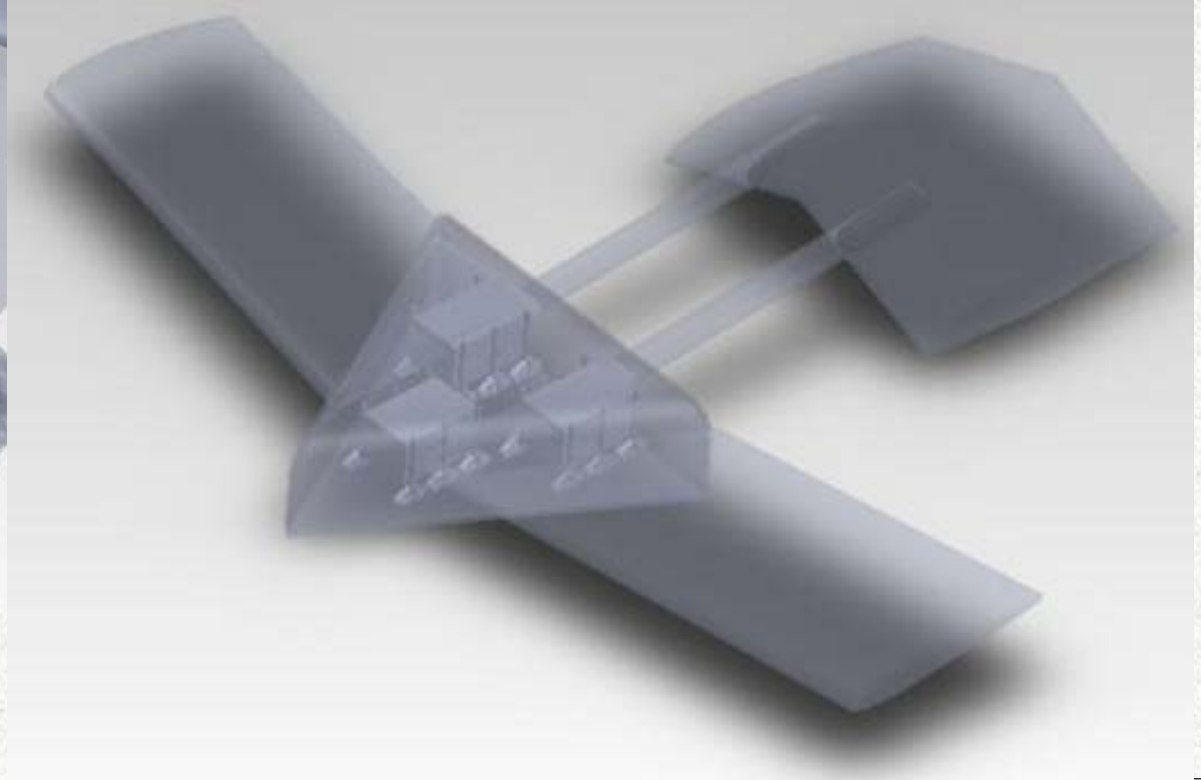
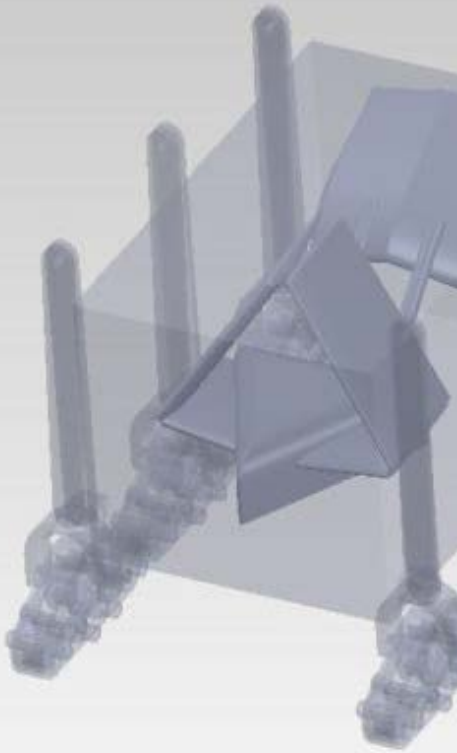
Terrain:

- Sand dunes leading up to, surrounding, and inside the chasm
- Rocky cliff faces $40^{\circ}+$ incline; landslide material $30-45^{\circ}$ incline
- Descent of 11km
- Rocky floor and roaming sand dunes

The mission:

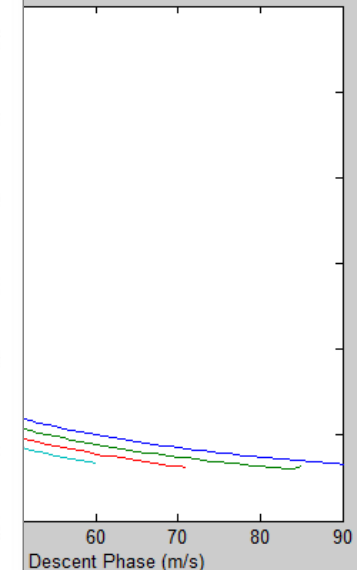
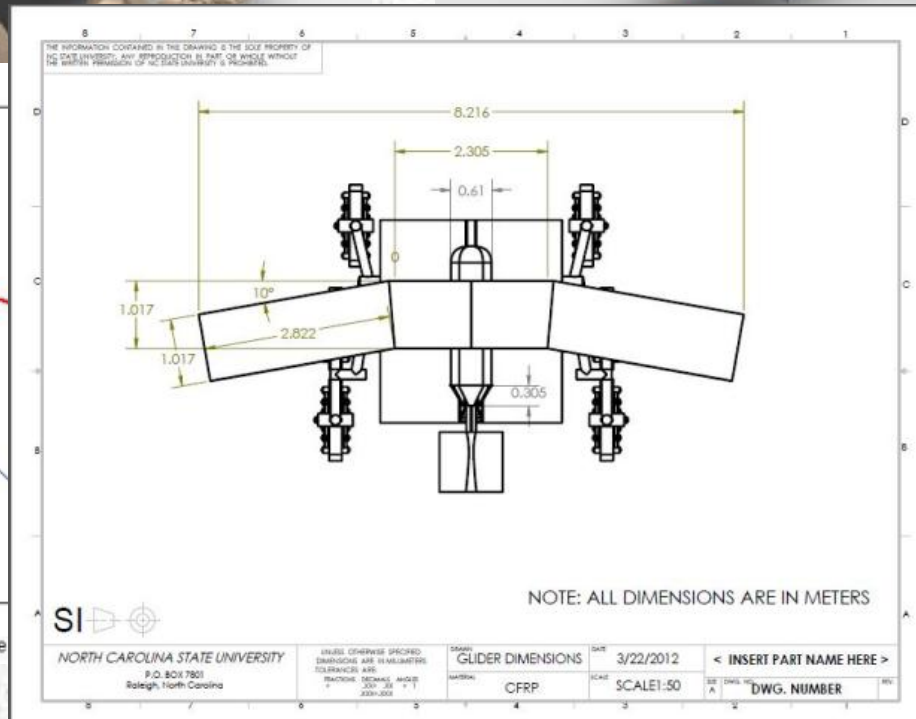
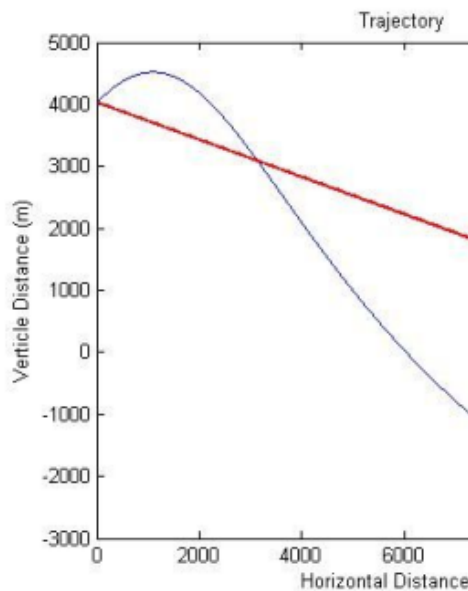
Find the most efficient way to land on the surrounding planes, traverse the dunes to reach the top of the chasm, and then descend to the floor while still being mobile once on the floor

Initial concept

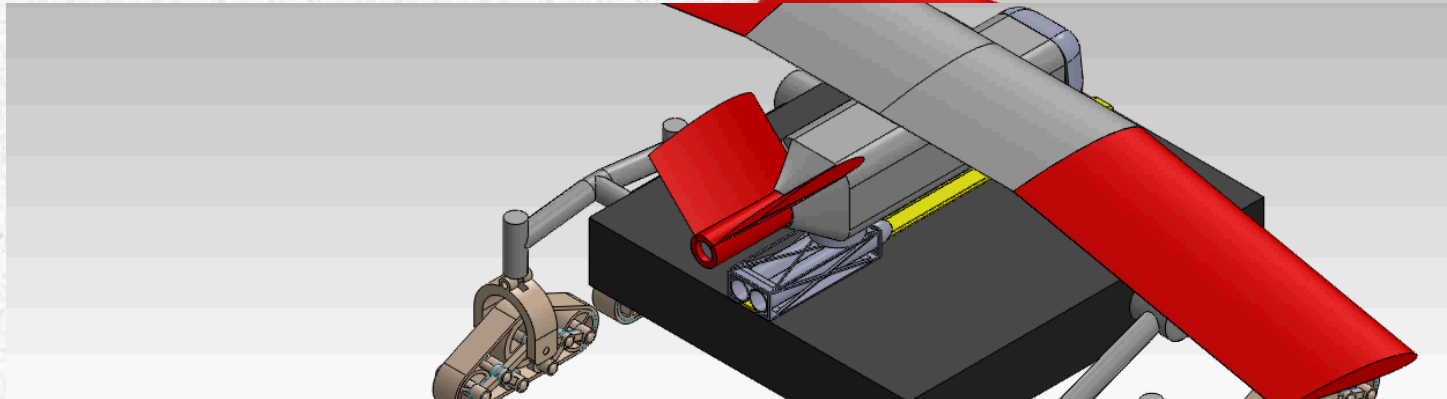
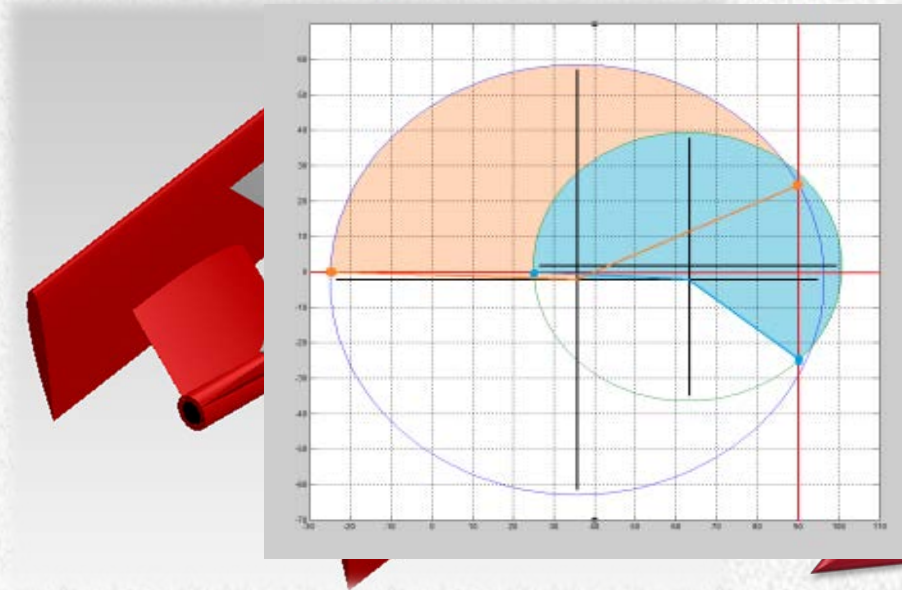


Designing the glider

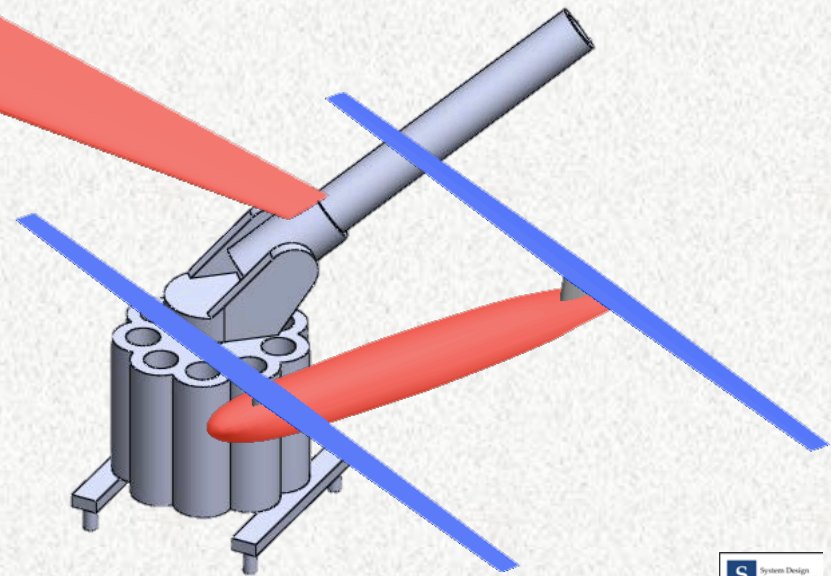
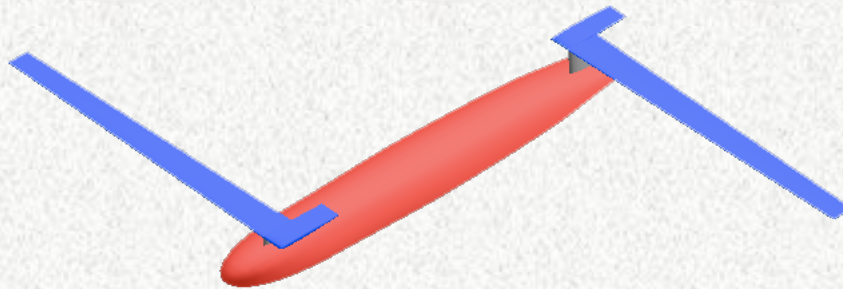
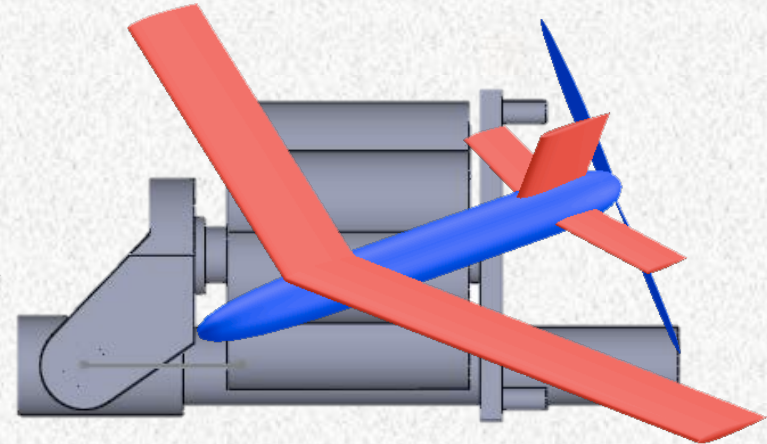
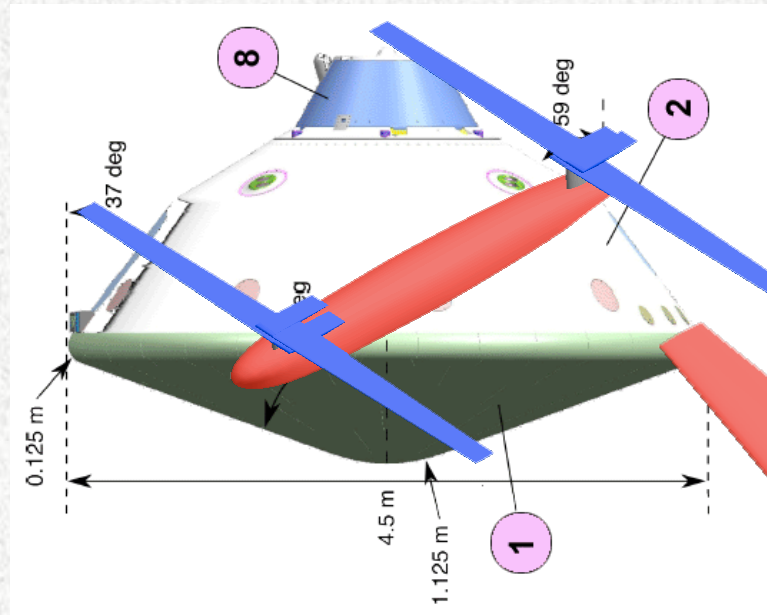
Material: Carbon Fiber Reinforced Polymer (CFRP)
Stall Speed: 100 m/s
Rocket Propellant (2 O-Motors)
Mass: 900 kg at Launch, 530 kg at Landing



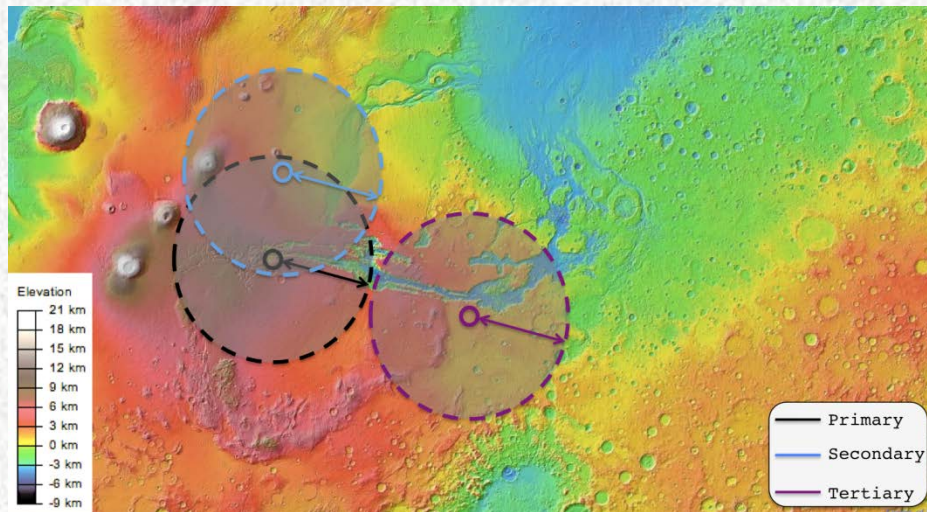
Describing the reconfigurations



Development of an air cannon



Requirements definition and analysis

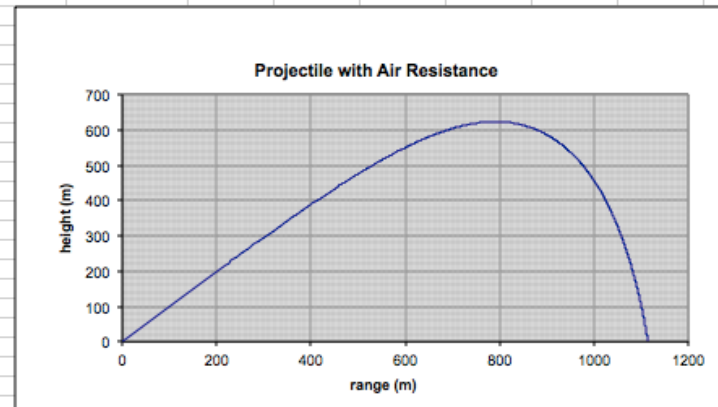


	A	B	C	D	E	F	G	H	I	J	K	L	M
1	Motion of a Projectile Within A Barrel												
2	Thermodynamic Closed System Analysis												
3													
4	mp =	15.00	kg										
5	L =	3.00	m										
6	D =	0.25	m										
7	P1 =	8000.00	kPa										
8	V1 =	4.00E+00	m^3										
9	T1 =	300.15	K										
10	Tc (CO2) =	304.19	K										
11	Pc (CO2) =	7380.00	kPa										
12	R =	8.31	J/mol-K										
13	Patm =	1.16	kPa										
14	MWco2 =	44.01	g/mol										
15	A =	0.05	m^2										
16	V2 =	4.15	m^3										
17	P2 =	7.72E+03	kPa										
18	TR1 =	0.99	--										
19	PR1 =	1.08	--										
20	Z1 =	1.00	--										
21	TR2 =	0.99	--										
22	PR2 =	1.05E+00	--										
23	Z2 =	1.00	--										
24	n =	1.28E+04	# moles										
25	mCO2 =	5.64E+05	g										
26	Wp =	1156929.08	J										
27	Wsurr =	170.09	J										
28	Wnet =	-1156759.00	J										
29	v_exit =	392.73	m/s										



ASSUMPTIONS: Quasi-Equilibrium Process
Gas behaves like and Ideal Gas (from compressibility EOS)
Gas expands isothermally
No heat exchanged
No change in internal energy
Neglect change in PE

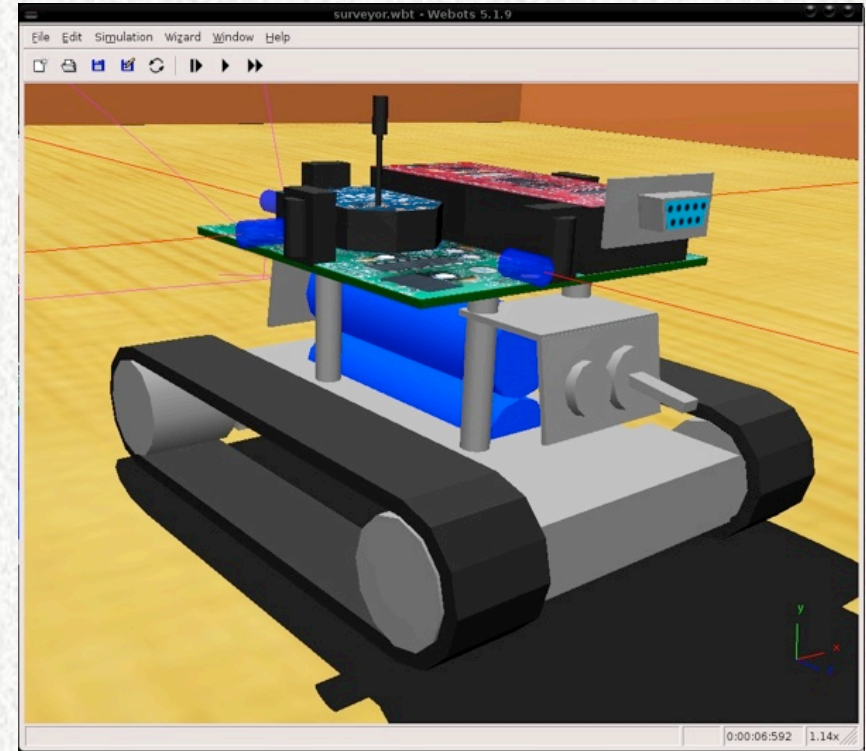
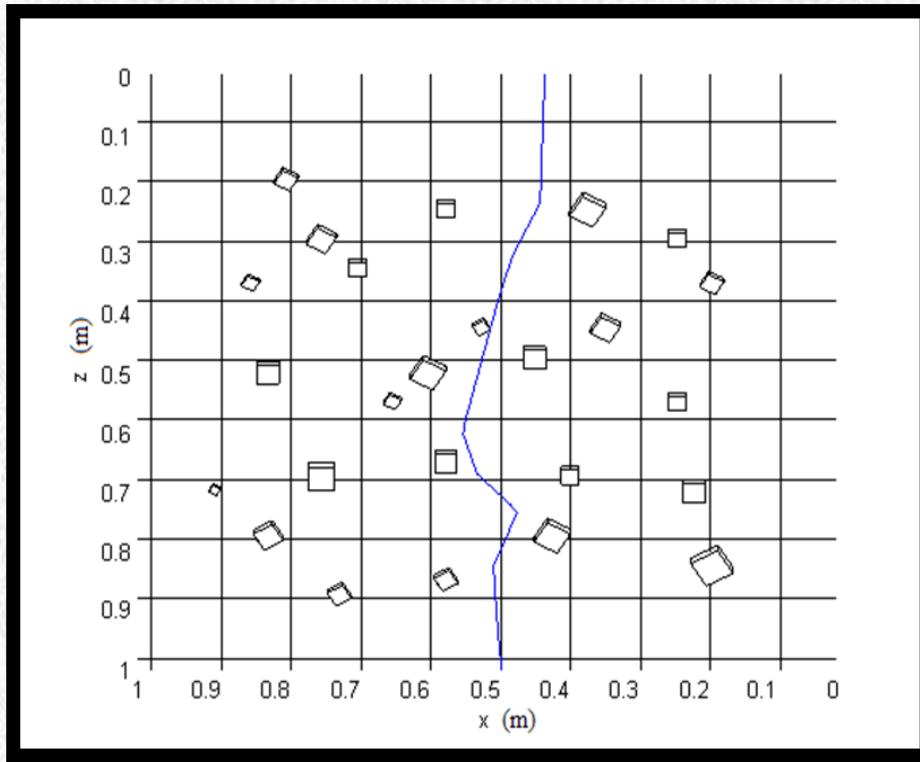
	A	B	C	D	E	F	G	H	I	J	K	L	M	N	O
1	Motion of a Projectile Under Gravity with Air Resistance														
2	Projectile fired at speed v at angle theta degrees to horizontal														
3	Air resistance = kv^2.														
4															
5	v =	392.73	m/s												
6	theta =	45.00	degrees	slider											
7	g =	3.71	m/s^2												
8	delta_t =	0.05	s												
9	k =	2.25E-03	--												
10	v_x_init =	277.70	m/s												
11	v_y_init =	277.70	m/s												
12	max range =	1114.01	m												
13	time @ max range =	36.60	s												
14	max height =	623.03	m												
15	range @ max height =	812.76	m												
16	time @ max height =	13.15	s												
17															
18	time	v_x	v_y	v	x	y									
19	0.00	277.7	277.7	392.72705	0	0									
20	0.05	265.4061	265.2206	375.20981	13.57765	13.57301									
21	0.10	254.1806	253.8175	359.20902	26.56732	26.54897									
22	0.15	243.8884	243.3545	344.53294	39.01905	38.97827									
23	0.20	234.4164	233.7177	331.02115	50.97667	50.90507									
24	0.25	225.6693	224.8112	318.53836	62.47881	62.36829									
25	0.30	217.5661	216.5533	306.96964	73.5597	73.4024									
26	0.35	210.0377	208.8744	296.21669	84.24979	84.03809									
27	0.40	203.0243	201.7144	286.19494	94.57634	94.30281									



Assumptions: No wave drag
k value chosen based on spherical geometry

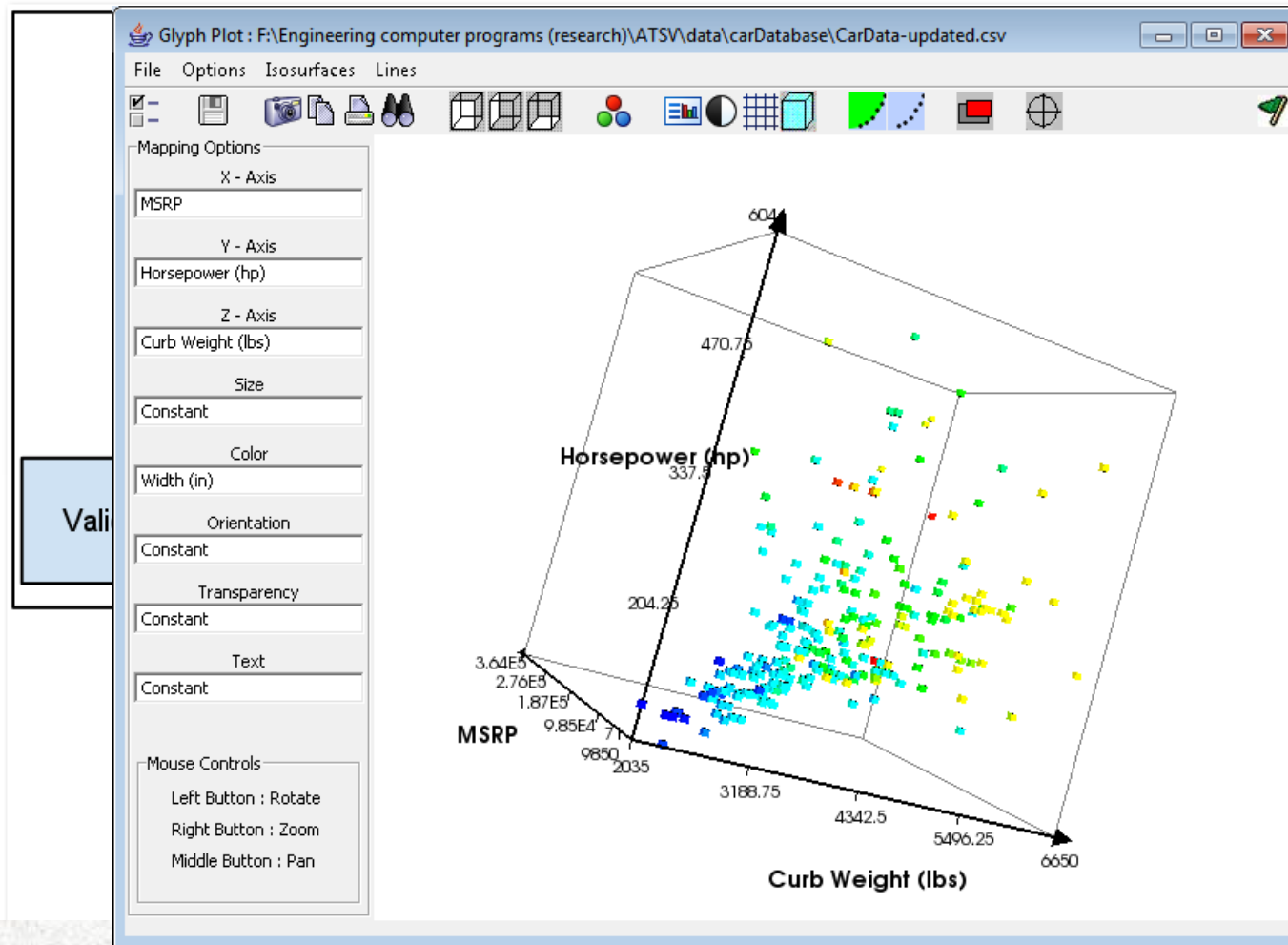
Where do we go from here?

- Advanced simulations
 - Webots simulation software

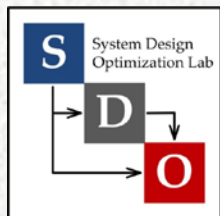


Where do we go from here?

- Tradespace exploration



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We thank NIAC for their support of this work

